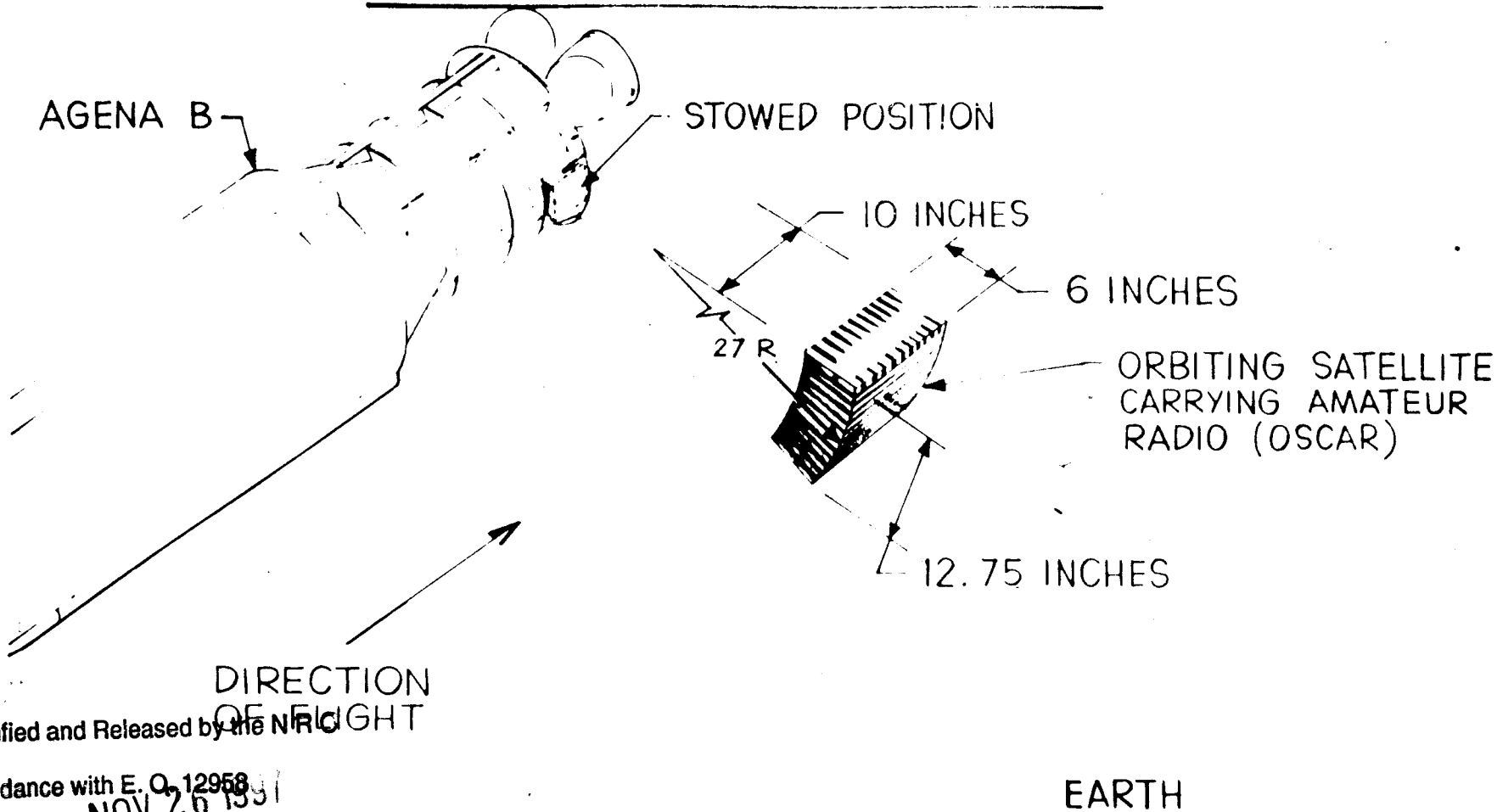


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DISCOVERER RESEARCH PROGRAM

EJECTION OF COMMUNICATION PACKAGE



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In Accordance with E. O. 12958

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PROPOSED GEOPHYSICAL EXPERIMENTS

	FTV NUMBER	1124	1125	1126	1132	1127	1128	1134	1129	1130	1131	1133	1135	1136
	DOOR NUMBER	112	113	114	115	116	117	118	119	120	121	122	123	124
1	EROSION DETECTOR													
1	IMPEDANCE PROBE 1A													
	TEMP. COMMUTATOR													
1	COSMIC RAY MON. 7B													
1	DENSITY GAGE MOD 3A													
1	RADIOMETER, MODEL 4													
1	RADIOMETER, MODEL 5													
1	CHARGED PARTICLE ENERGY													
1	COSMIC RAY MONITOR 8B													
2	GALACTIC RF MON, 3													
1	ELECTRON DENSITY GAGE 2C													
3	ULTRA VIOLET RADIOMETER													
2	GALACTIC RF MON 2													
	RETARD POTENT. ANALYZER													
1	ION TRAP													
	BETA-GAMMA DETECTOR													
	TISSUE EQUIVALENT CHAMBER													
2	IMPEDANCE PROBE 2													
1	MAGNETOMETER													
2	20/40 TRANSMITTER													
	NEUTRON DETECTOR 3													
	MASS SPECTROMETER													
	RADIOMETER, MODEL 6													
	MICKROMETEORITE DETECTOR, GRAV-ACOUSTIC													
	1 EXPERIMENT MOUNTED ON ERECTABLE BOOM													
	TWO TAPE ANTENNAE REQUIRED													
	SENSORS FACING EARTH													
	SENSOR FACING ZENITH													

VELA
HOTEL

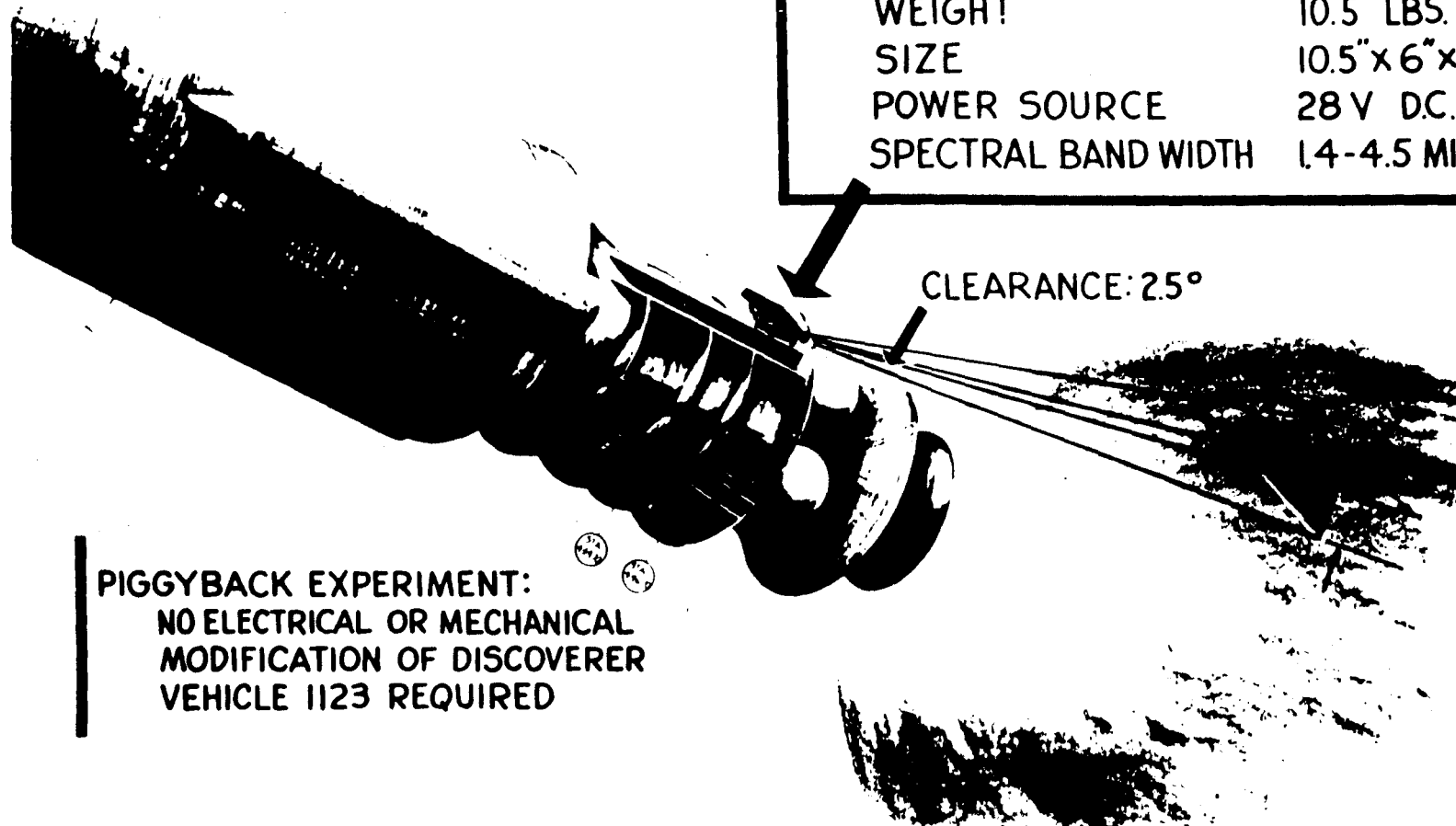
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MIDAS PLUME

SPECTROMETER CHARACTERISTICS:

WEIGHT	10.5 LBS.
SIZE	10.5" x 6" x 4.6"
POWER SOURCE	28 V DC.
SPECTRAL BAND WIDTH	1.4-4.5 MICRONS



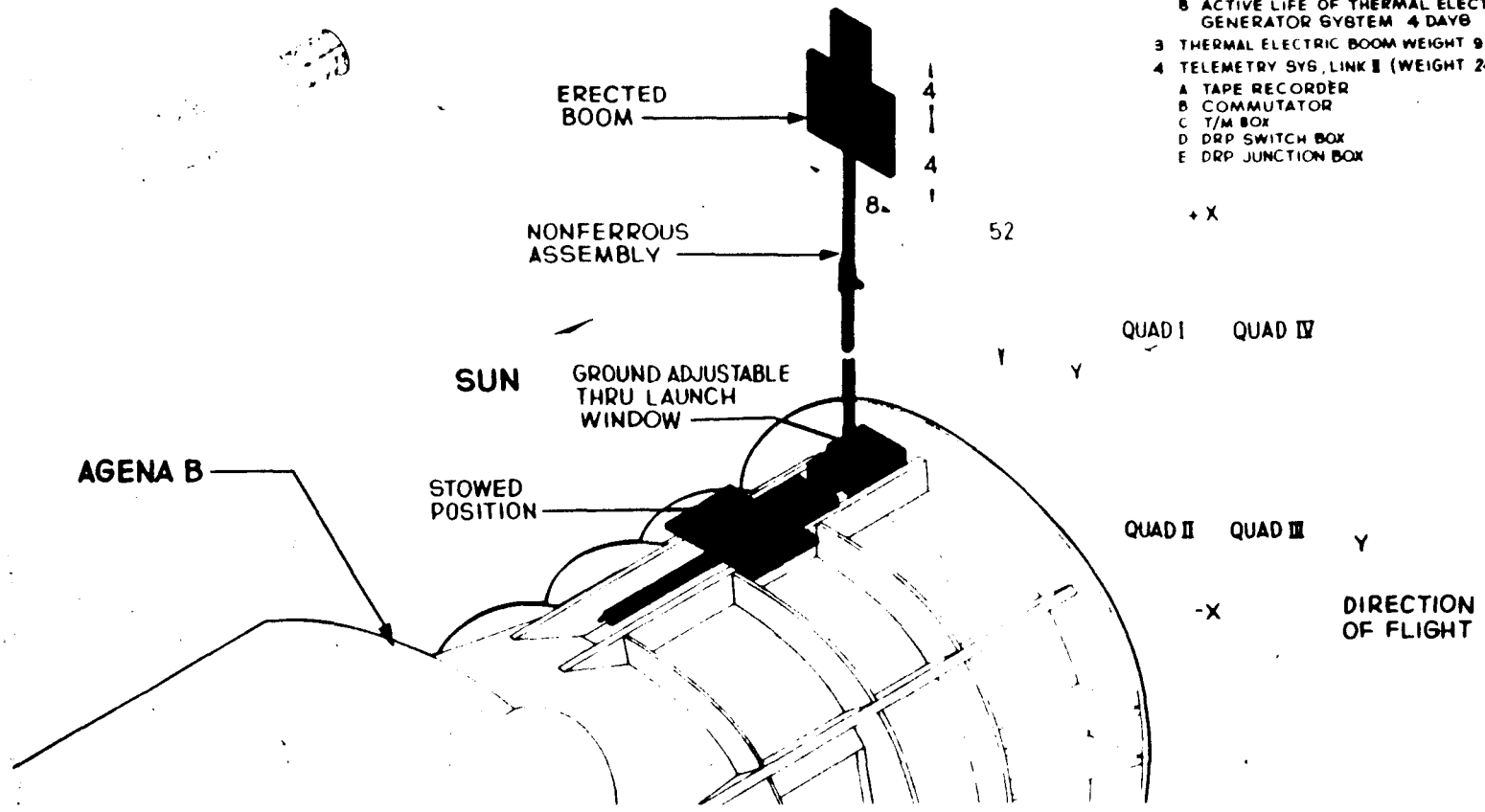
CLEARANCE: 2.5°

**PIGGYBACK EXPERIMENT:
NO ELECTRICAL OR MECHANICAL
MODIFICATION OF DISCOVERER
VEHICLE 1123 REQUIRED**

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THERMOELECTRIC BOOM



NOTES

- 1 TEST OBJECTIVE DETERMINE THE OPERATING CHARACTERISTICS OF A THERMOELECTRIC GENERATOR IN SPACE ENVIRONMENT
- 2 OPERATION
 - A THERMOELECTRIC GENERATOR EXTEND IN PLACE AFTER VEHICLE PLACED IN ORBIT
 - B ACTIVE LIFE OF THERMAL ELECTRIC GENERATOR SYSTEM 4 DAYS
- 3 THERMAL ELECTRIC BOOM WEIGHT 90 LBS
- 4 TELEMETRY SYS, LINK II (WEIGHT 24 LBS)
 - A TAPE RECORDER
 - B COMMUTATOR
 - C T/M BOX
 - D DRP SWITCH BOX
 - E DRP JUNCTION BOX

+ X

QUAD I QUAD IV

QUAD II QUAD III

-X

DIRECTION OF FLIGHT

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HORIZON SENSOR EXPERIMENT

OBJECTIVES:

TO EVALUATE THE PERFORMANCE
OF A HORIZON SENSOR FOR
CONTROL OF VEHICLE ORIENTATION

EQUIPMENT:

2 HORIZON SCANNERS
1 MIXER BOX
1 PRE-AMP. BOX
TOTAL EQUIP. WT. 13.5 LBS.

TELEMETRY:

7 INSTRUMENTATION POINTS WITH 2 ON CONT CH
REAL TIME DATA VIA LINK II T/M
POWER REQUIREMENTS, 12 WATTS

HORIZON SENSOR



MIXER BOX

HORIZON SENSOR

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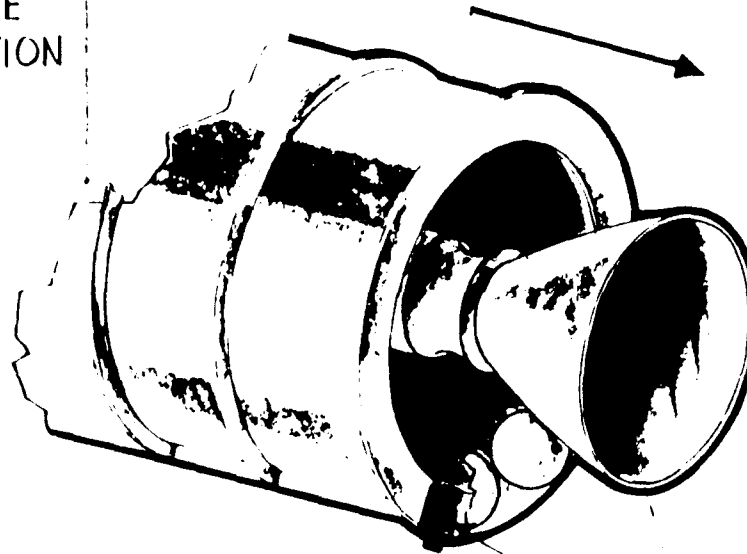
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ULTRA VIOLET RADIOMETER EXPERIMENT

OBJECTIVE:
TO DETERMINE THE
OZONE DISTRIBUTION
AS A FUNCTION
OF ALTITUDE

DIRECTION OF FLIGHT



COMMUNICATIONS:
• 4 INSTRUMENTATION POINTS
• REAL TIME DATA VIA
LINK II T/M
• RECORDED DATA ON
DUAL TRACK 180 MIN.
TAPE RECORDER
(27:1 RATIO).

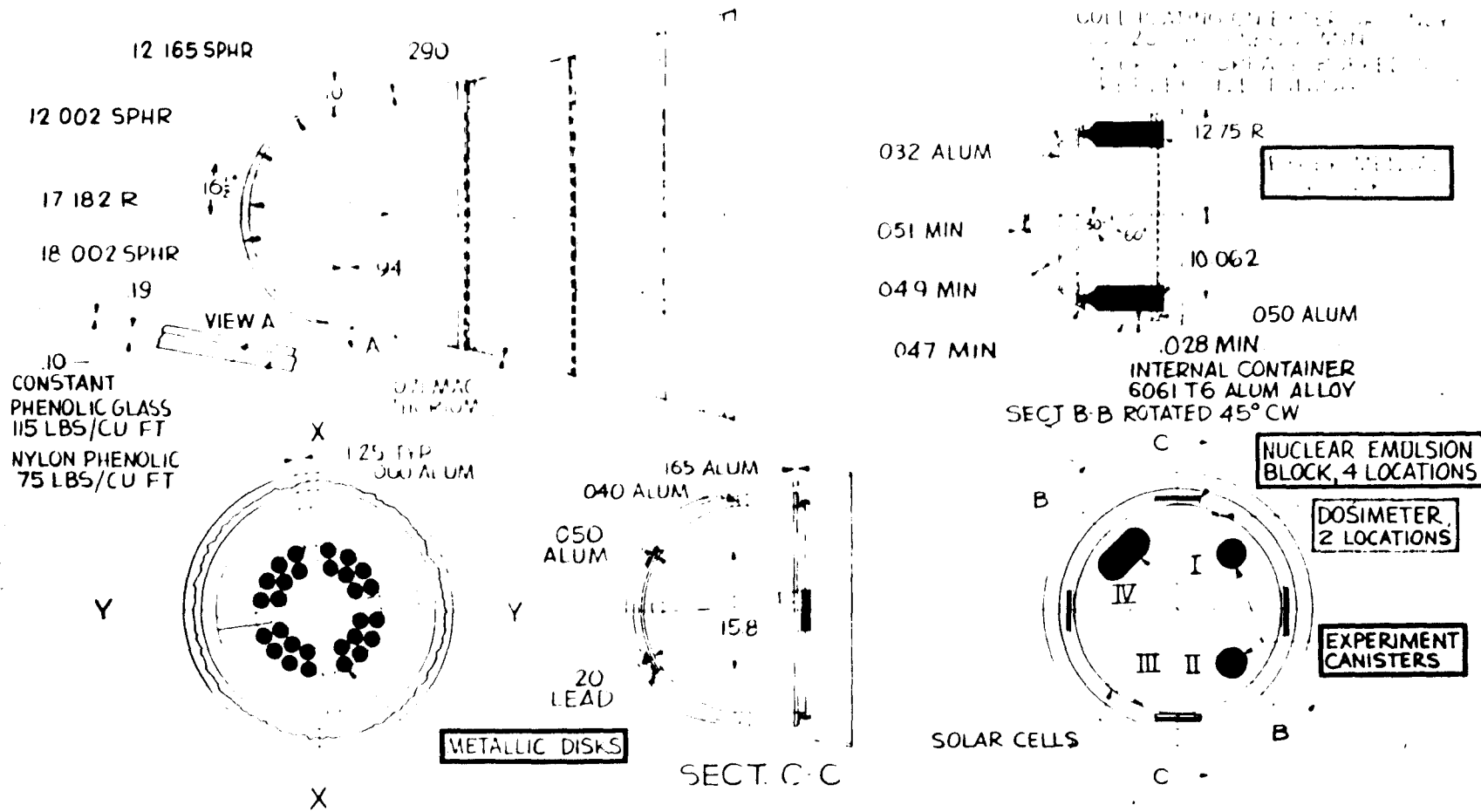
UV RADIOMETER

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DISCOVERER RECOVERY CAPSULE



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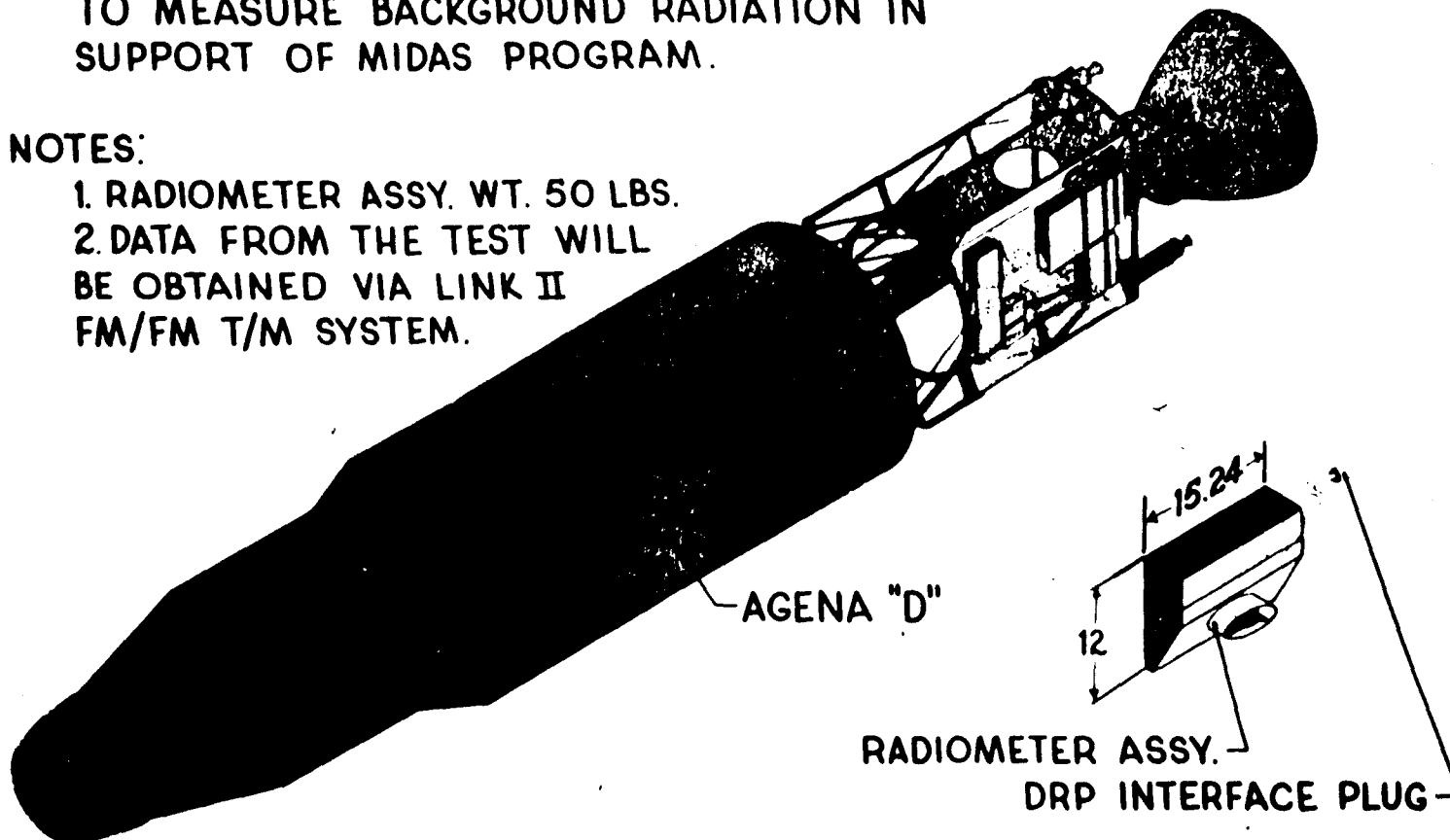
RADIOMETER TEST

OBJECTIVE:

TO MEASURE BACKGROUND RADIATION IN
SUPPORT OF MIDAS PROGRAM.

NOTES:

1. RADIOMETER ASSY. WT. 50 LBS.
2. DATA FROM THE TEST WILL
BE OBTAINED VIA LINK II
FM/FM T/M SYSTEM.



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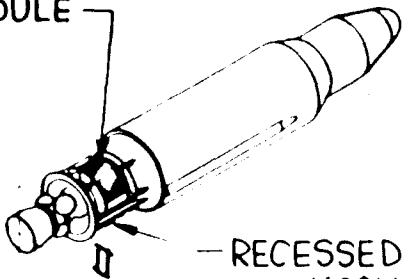
DISCOVERER RESEARCH PROGRAM

RECESSED & LARGE DOOR MODULES

NOTES:

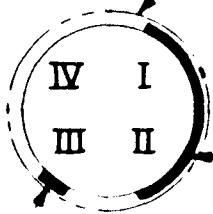
- TEST OBJECTIVES: TO PROVIDE A BASIC CAPABILITY FOR CONDUCTING SPACE ENVIRONMENT AND MATERIAL EXPERIMENTS
- MODULE WEIGHT
LARGE DOOR WITH BASIC EQUIP. - 35 LBS
RECESSED DOOR, VARIABLE
- ACTIVE LIFE OF MODULES - 4 DAYS
- BASIC EQUIP ON LARGE DOOR MODULE
TAPE RECORDER
PROGRAMMER
TWO COMMUTATORS
TIME REFERENCE GENERATOR
DC/DC POWER SUPPLY

LARGE DOOR MODULE



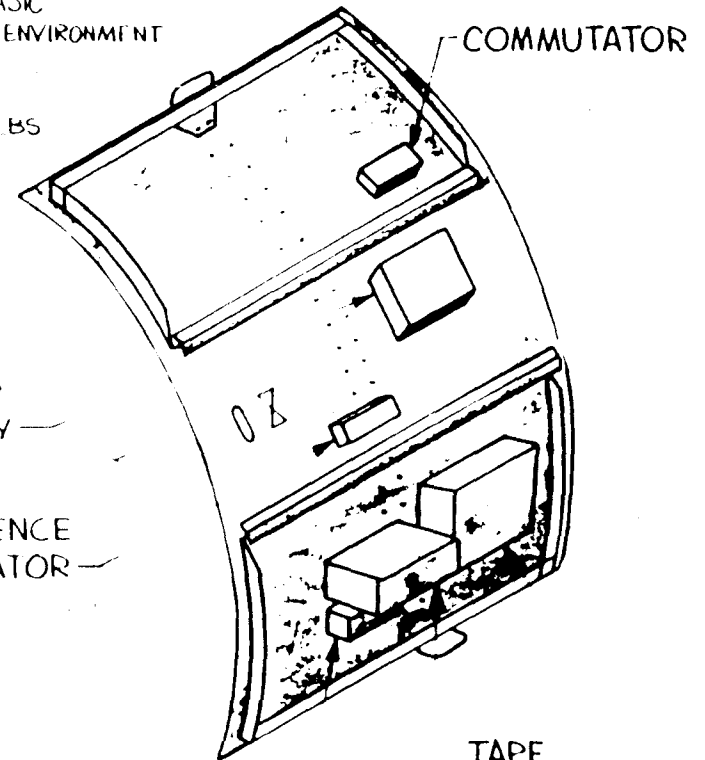
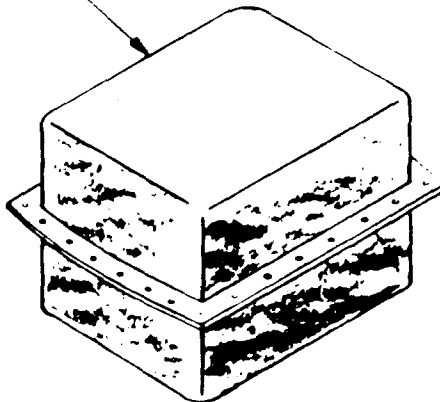
RECESSED DOOR MODULE

THOR INTERFACE



LARGE DOOR MODULE

RECESSED DOOR MODULE



COMMUTATOR

PROGRAMMER

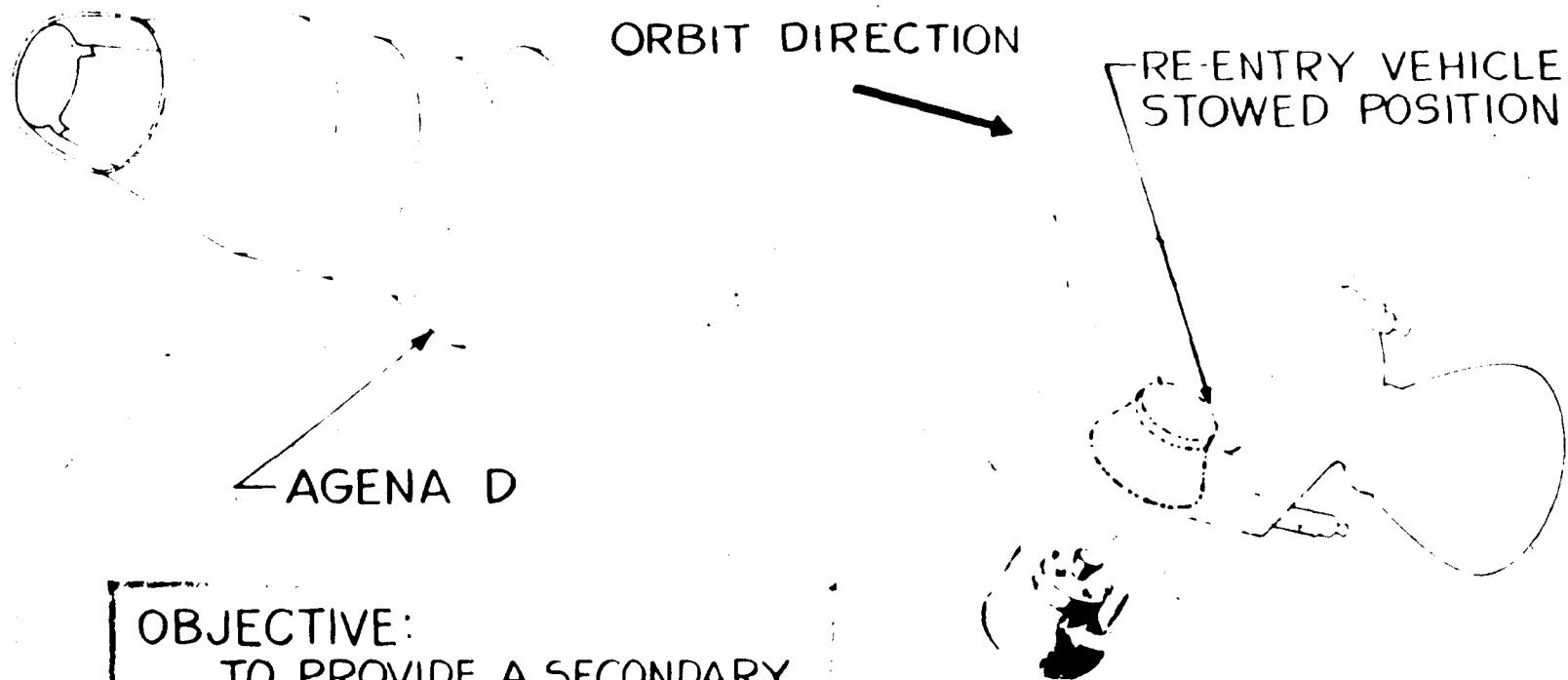
LARGE DOOR MODULE

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RE-ENTRY RESEARCH VEHICLE

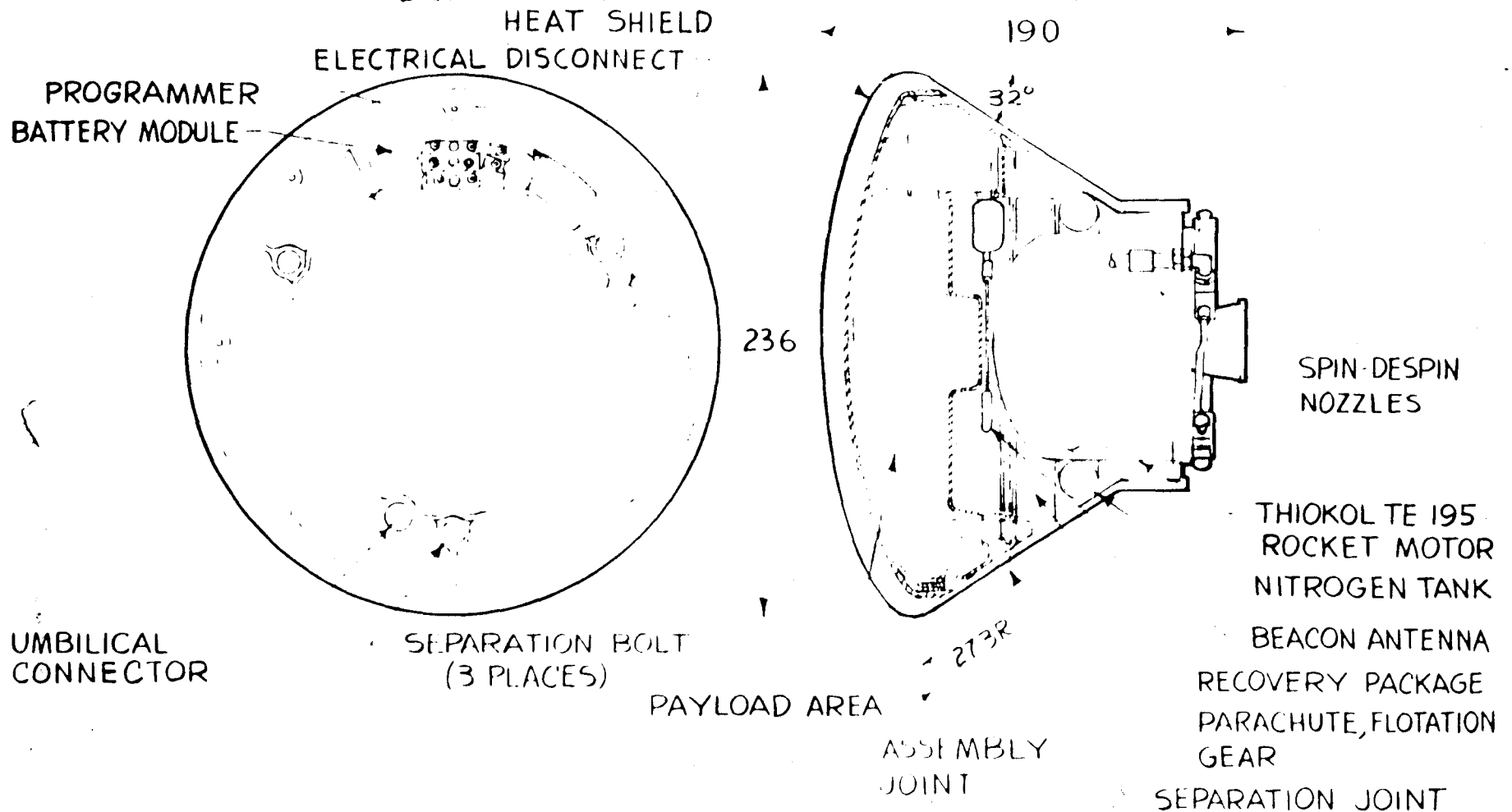


OBJECTIVE:
TO PROVIDE A SECONDARY
RE-ENTRY VEHICLE FOR
RESEARCH EXPERIMENTS

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RE-SEARCH RE-ENTRY VEHICLE



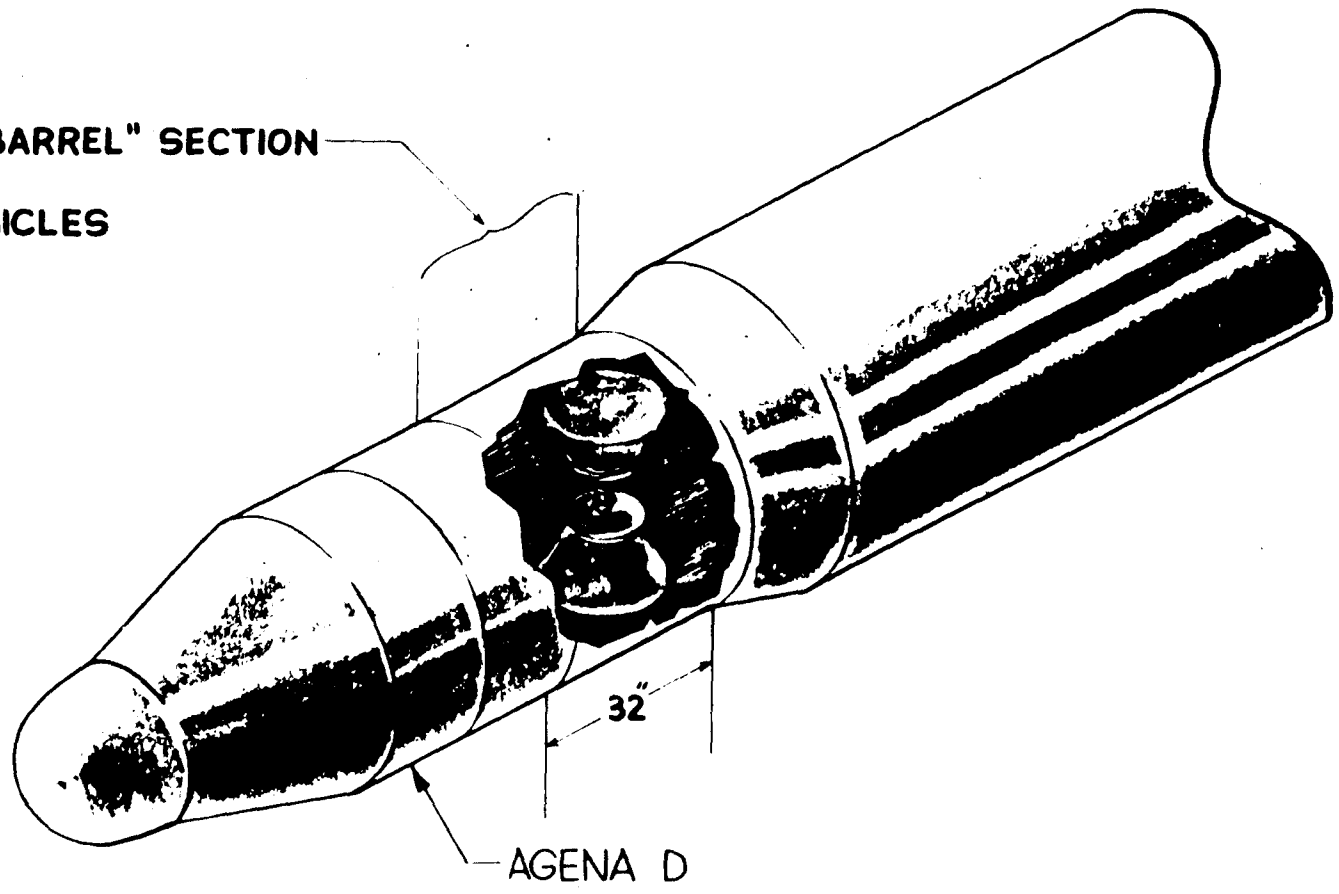
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ALTERNATE STOWAGE CONCEPT OF RE-ENTRY RESEARCH VEHICLES

ADDITIONAL "BARREL" SECTION
STOWS TWO
RE-ENTRY VEHICLES



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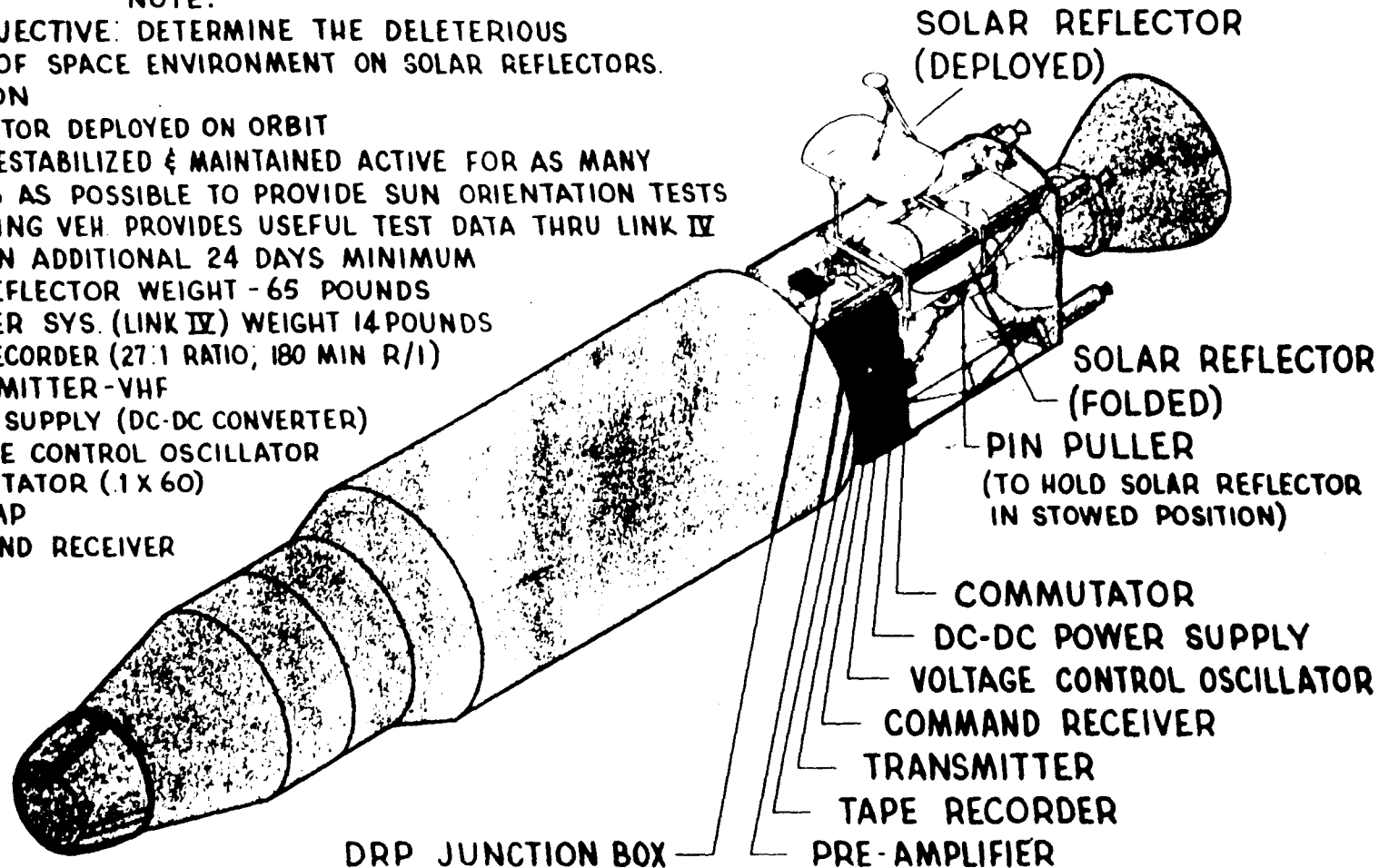
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SOLAR REFLECTOR SYSTEM INSTALLATION

DISCOVERER

NOTE:

1. TEST OBJECTIVE: DETERMINE THE DELETERIOUS EFFECTS OF SPACE ENVIRONMENT ON SOLAR REFLECTORS.
2. OPERATION
 - A. REFLECTOR DEPLOYED ON ORBIT
 - B. VEH. RESTABILIZED & MAINTAINED ACTIVE FOR AS MANY ORBITS AS POSSIBLE TO PROVIDE SUN ORIENTATION TESTS
 - C. TUMBLING VEH. PROVIDES USEFUL TEST DATA THRU LINK IV FOR AN ADDITIONAL 24 DAYS MINIMUM
3. SOLAR REFLECTOR WEIGHT - 65 POUNDS
4. TELEMETER SYS. (LINK IV) WEIGHT 14 POUNDS
 - A. TAPE RECORDER (27:1 RATIO, 180 MIN R/I)
 - B. TRANSMITTER - VHF
 - C. POWER SUPPLY (DC-DC CONVERTER)
 - D. VOLTAGE CONTROL OSCILLATOR
 - E. COMMUTATOR (1 X 60)
 - F. PRE-AMP
 - G. COMMAND RECEIVER



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